









# **Nondestructive Testing of Fracture and Failure Critical Components**

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50th Annual Air Transportation Association (ATA) Non-Destructive Testing (NT) Forum  
Hyatt Regency Hotel, Orlando International Airport  
Orlando, Florida  
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# OUTLINE

- **Columbia Mishap**
- **NASA NDE Team & Agency NDE Program**
- **Space Transportation System Inspection (Advanced NDI)**
- **On-Orbit NDT (Inspectors in Space)**
- **NASA Core NDT Standards**
- **Looking Toward the Future**
- **Summary**



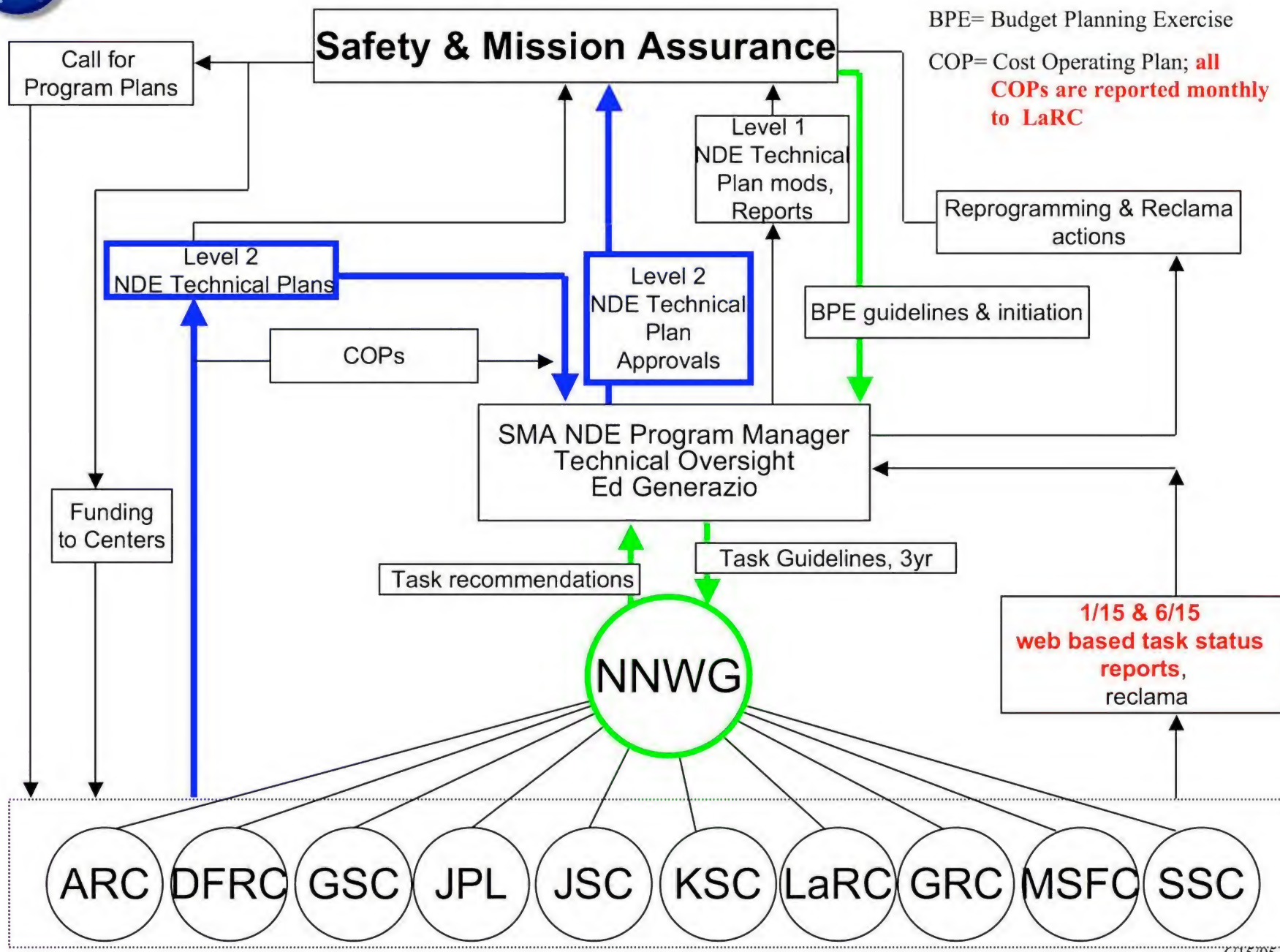
- Launch Movie
- Debris Strike Movie
- Foam Impact on RCC – Outside View
- Foam Impact on RCC – Inside View



# External Tank TPS









# Agency NDE Program

- Physical Brittle Matrix Composites Standards for validating inspection systems.
- Scanning materials characterization System applied to impact and oxidation damaged Orbiter RCC Leading Edge.
- Enhanced Micro-focus X-ray Tomography for applications to Orbiter RCC, CEV seals, and metallic sandwich foams.
- NDE Capability Assessment (thermal, ultrasonic, eddy current, x-ray tomography, backscatter X-ray, for characterization of impact and oxidation damaged Orbiter RCC Leading Edge.
- NDE Capability Assessment of Laser Vibrometer for global damage characterization in structures
- NDE Capability Assessment of Air Coupled Ultrasonics for bond evaluation of satellite systems (mirror segments, etc.)
- NDE Capability Assessment of Time Domain Reflectometry for Wiring Integritiy (Vibration test-bed included: bend radii, sharp corner, oversized clamp, clamp missing rubber, exposed conductor, fraying, missing clamp, chaffing.
- NDE Capabilities Assessment of Electric Field Imaging for wire integrity.
- NDE Standard Guide (WK8211) and Practices [*Acoustic Emissions (WK12759)*, *Thermal (WK12737)*, *Shearography (WK12796)*, *Radiography (13191)*, *Ultrasonics (11990)*] for Composites
- NDE Technology Assessment (Acoustic Emission and Thermography) supporting 40" OMS COPV test.
- NDE Technology Assessment for COPVs, and 100 aged Physical COPV Standards for characterizing COPV age and state of stress rupture.
- Advanced NDE Technology for High Performance COPV
- Eddy Current On-Orbit Inspection System and Physical Standards for and Intravehicular activity (IVA) characterization of micrometeoroid damaged to Space Station skin panels.
- NDE Capabilities Assessment of thermography, backscatter x-ray, and millimeter wave, terahertz inspection technology for corrosion under paint, primer, and RTV Silicon (Orbiter configuration), Tile Bond, Gap Filler Impregnation Level, FRSI, and SRB, supporting International Space Station (ISS), Constellation (Cx), Launch Services (LSP), Space Shuttle (SSP) Programs, and aging Ground Support Equipment (GSE).
- NDE Technology for ET close-outs and crushed foam at KSC, including: 3<sup>rd</sup> hardpoint, Orbiter Jack Panel and Umbilical Pyro Bolt, and Upper and Lower SRB Attach Points.
- NDE Technology for Orbiter Tile Bond Verification
- Computational Simulations of On-Orbit Solar Heating Thermography for TPS integrity
- Standard Methods for Generating Cork, Foam, and Carbon Fiber Physical Standards for NDE capability assessments for CLV, ET, RSRM, SRB, and Ares I Programs
- Health Monitoring identification of load and active areas of load distribution during COPV burst testing for SSP.
- Materials Engineering Properties at millimeter wavelengths for NDE systems calibrations applied to ET TPS inspections.
- NDE Capability Assessment of Limited Angle Reverse Geometry X-ray Laminography for Double Wall (12' apart) ET SOFI inspection, RCC, and Orbiter TPS Tile.

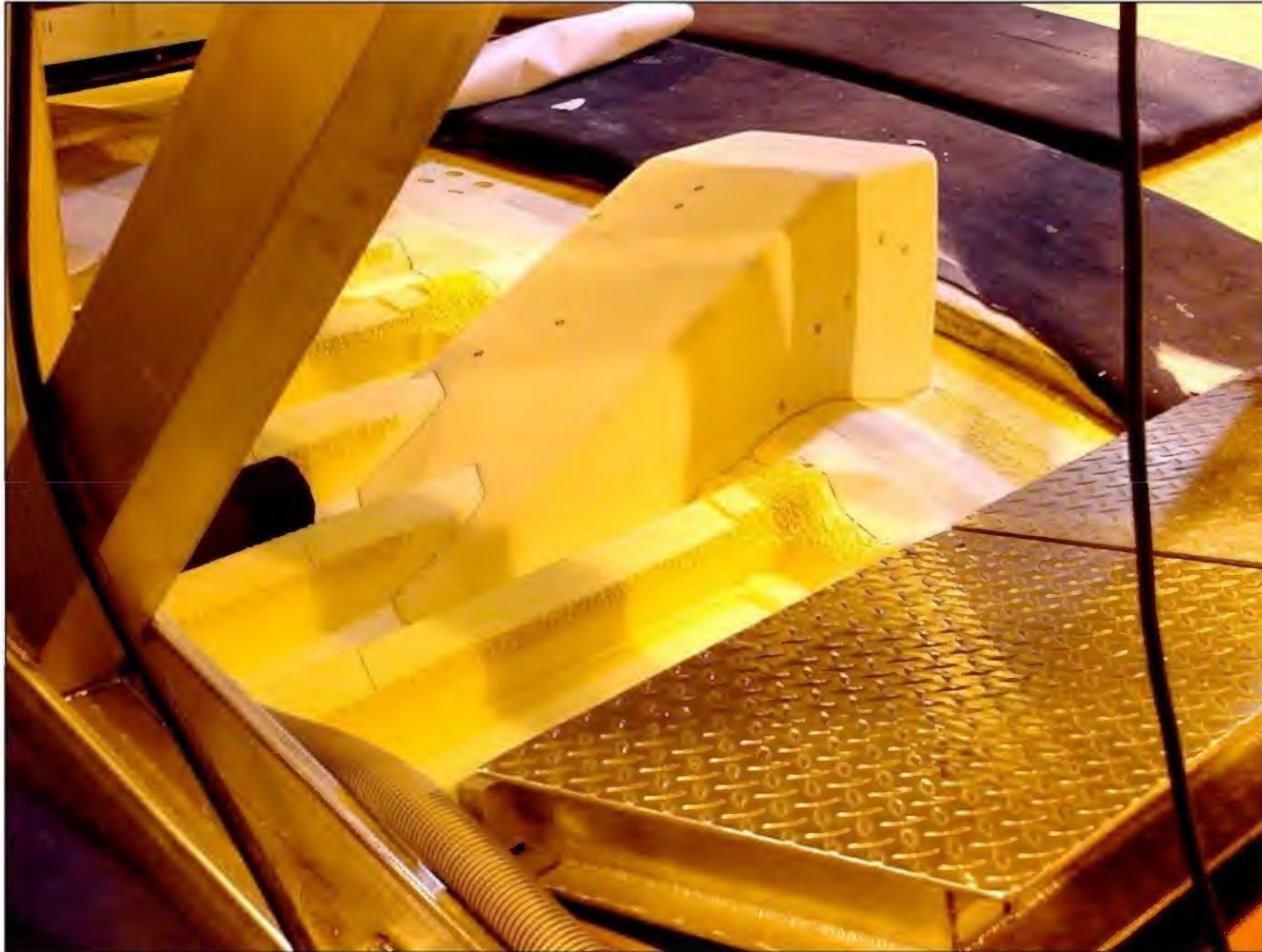




# Terahertz NDE



## A View of the SOFI Closeout Region



External Tank SOFI closeout region



## Bipod Joint Structure Under SOFI

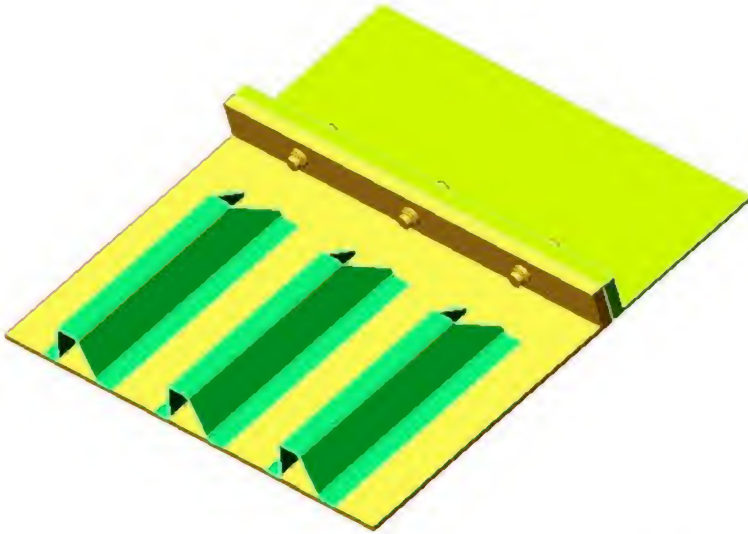


NDE mockup of bipod joint with SOFI removed.



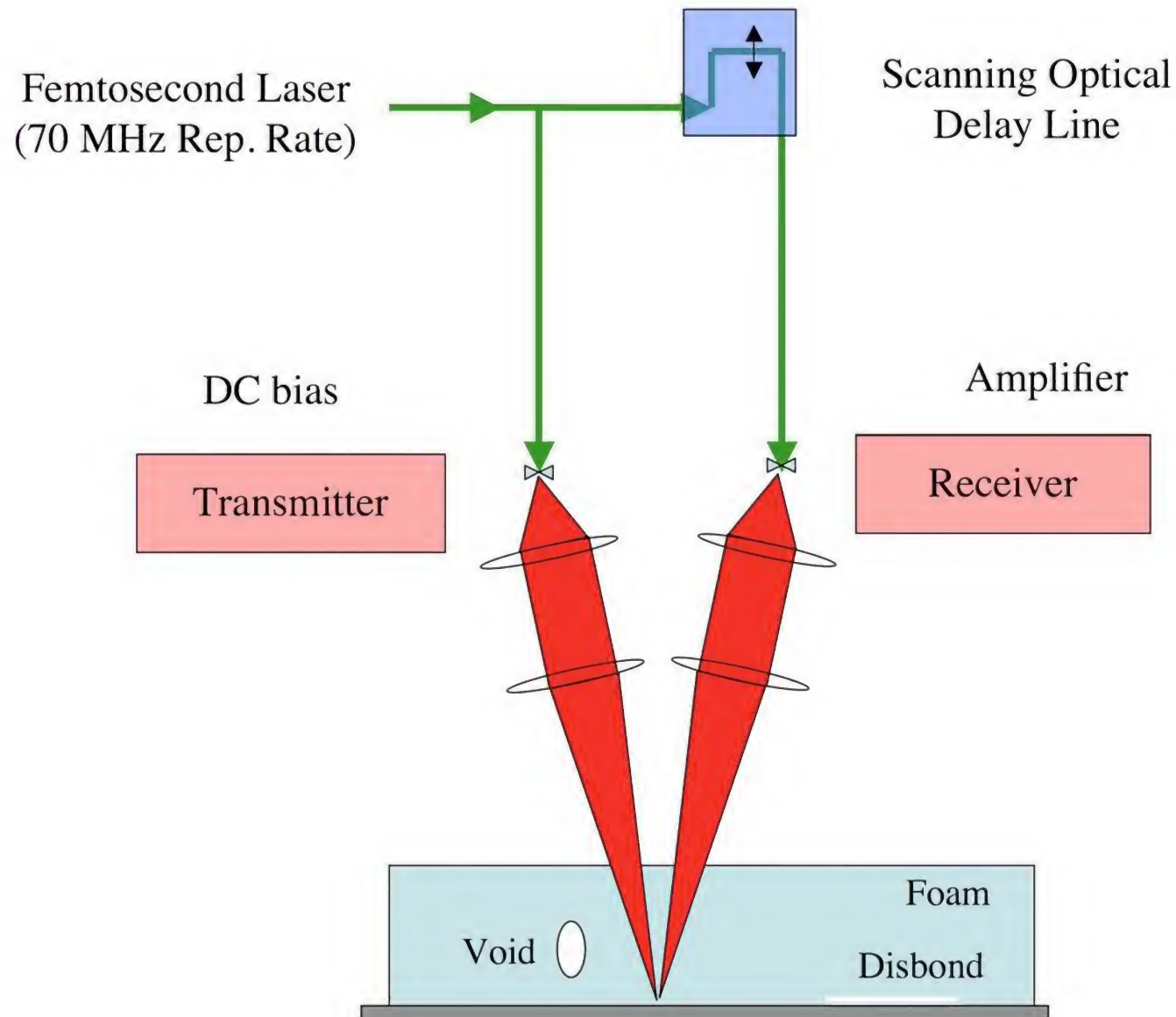


## Probability of Detection (POD) Foam Panel Model





# Time Domain Terahertz System

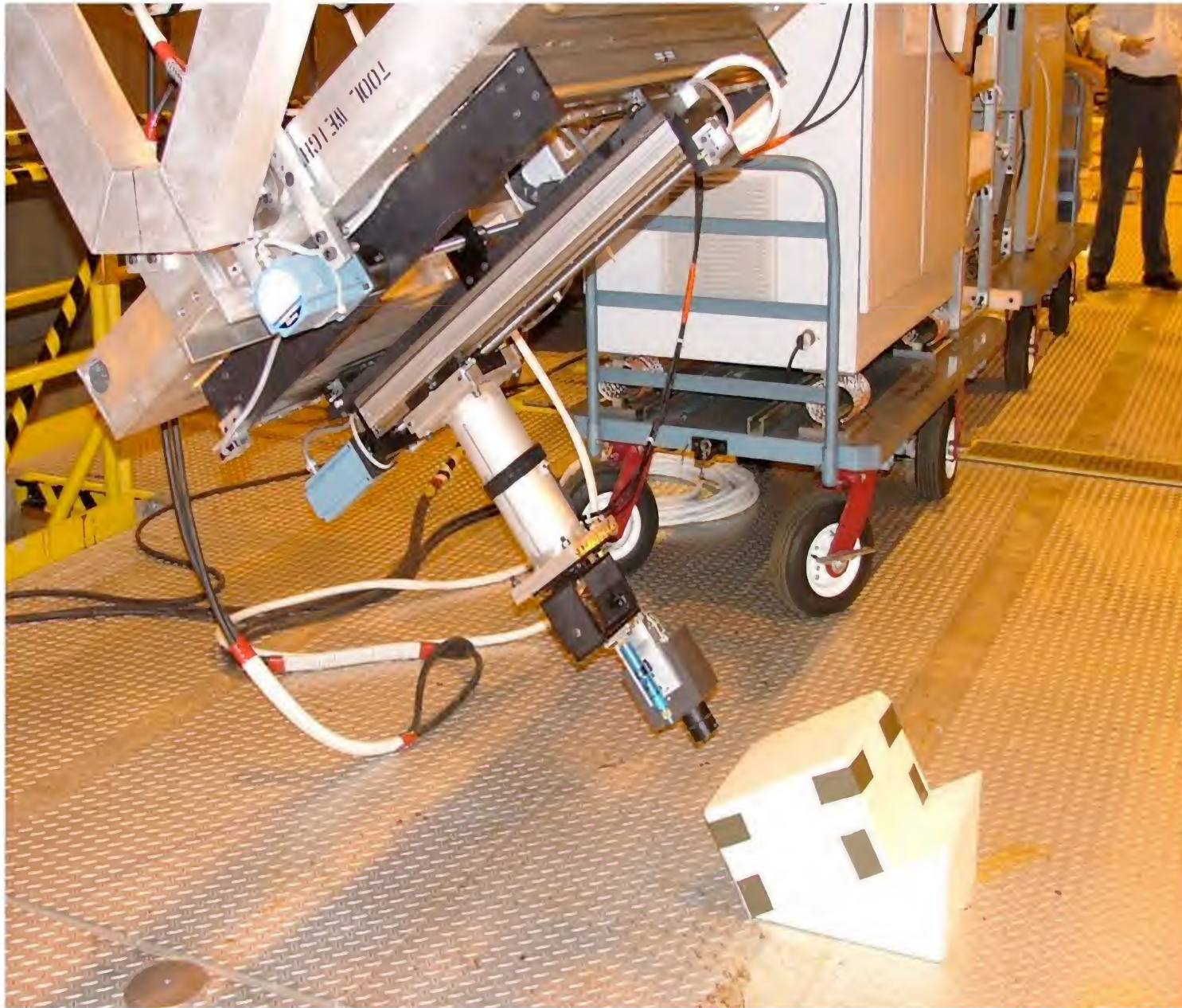






## Terahertz SOFI Inspection System

14





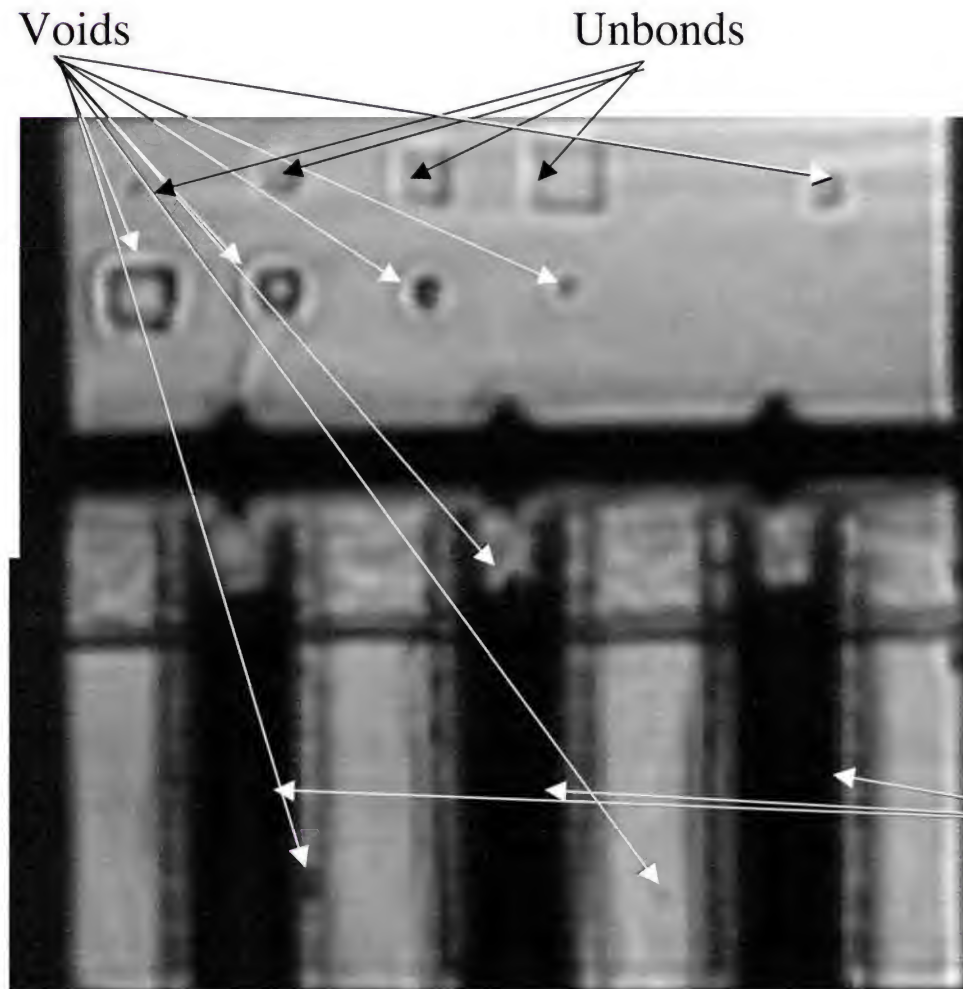


# Backscatter X-Ray SOFI Inspection System





# Terahertz imaging of SOFI with flaws



Terahertz image of flaws in a test panel with complex geometry. This figure was made with LaRC's new T-ray system.

Flange joint under 2" of SOFI

Stiffeners under 2" of SOFI

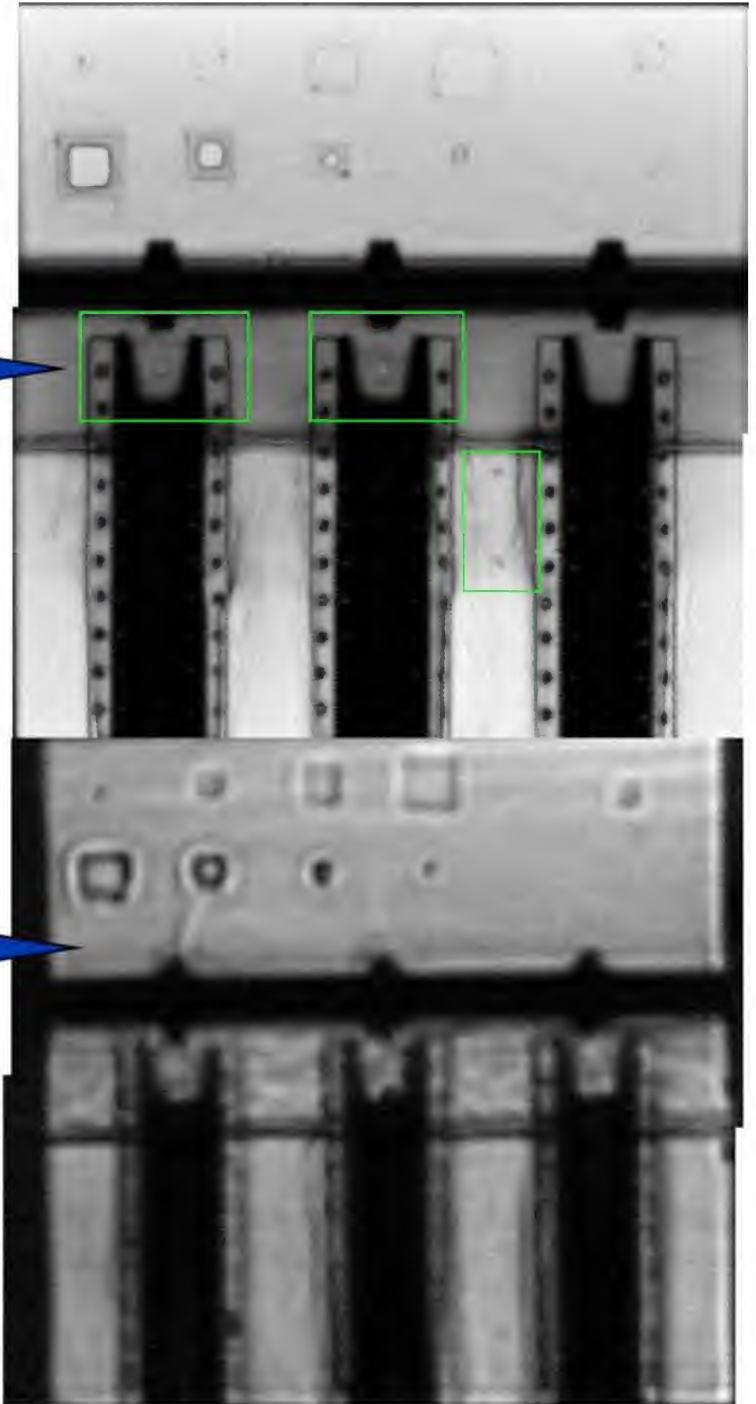
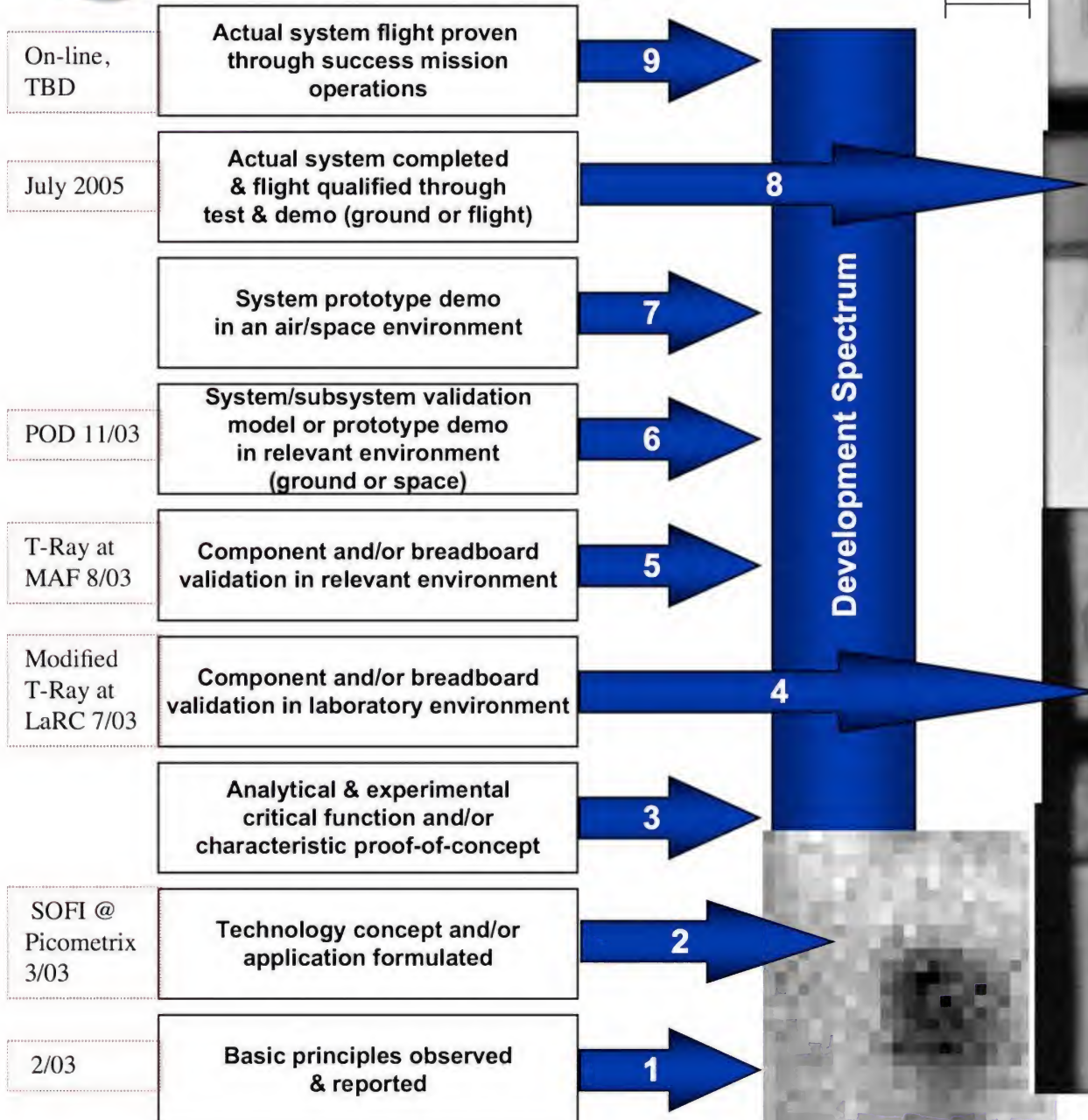




## NASA Technology

## Terahertz NDE

### Readiness Level (TRL)



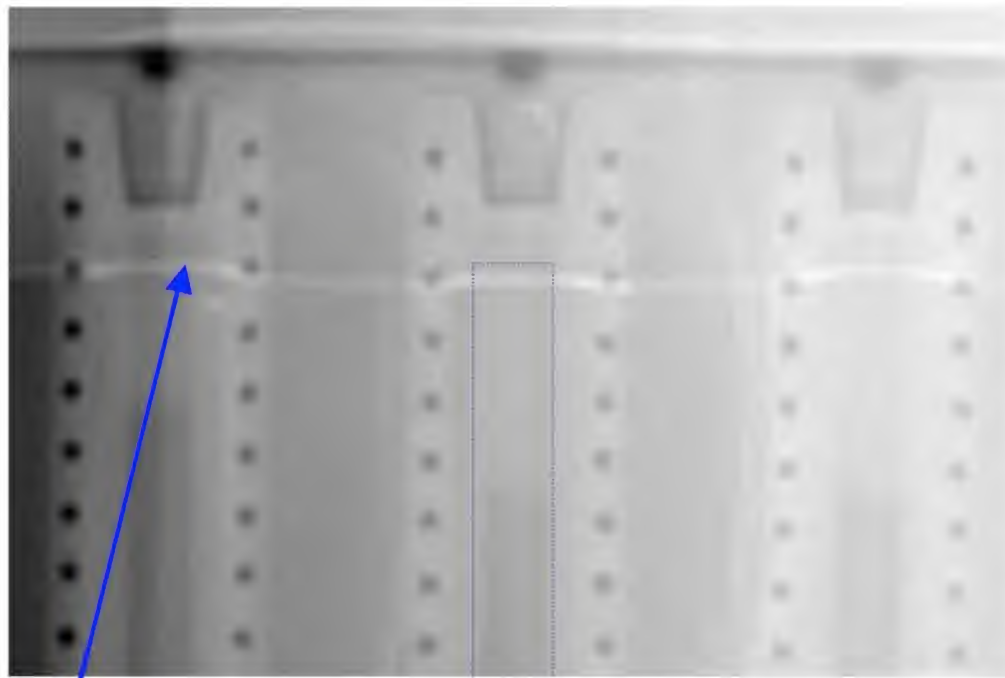




# Backscatter X-ray



## SOFI Backscatter X-ray



Enhanced image shows:

Filled region

Empty region

conathane

- Backscatter X-ray results, 2" of SOFI



## Non-Destructive Evaluation (NDE) for TPS

(March 2005)

Certification Status: Micro-focus x-ray will be certified for engineering evaluations on stringer closeouts in April 2005. THZ and BSX will be certified for inspection of LH2 stringer closeouts and PAL ramps in August 2005.

(August 2005)

Certification Status: Micro-focus x-ray, THZ and BSX will be certified for engineering evaluations on stringer closeouts in August 2005. THZ and BSX will be certified for inspection of LH2 stringer closeouts and PAL ramps in December 2005

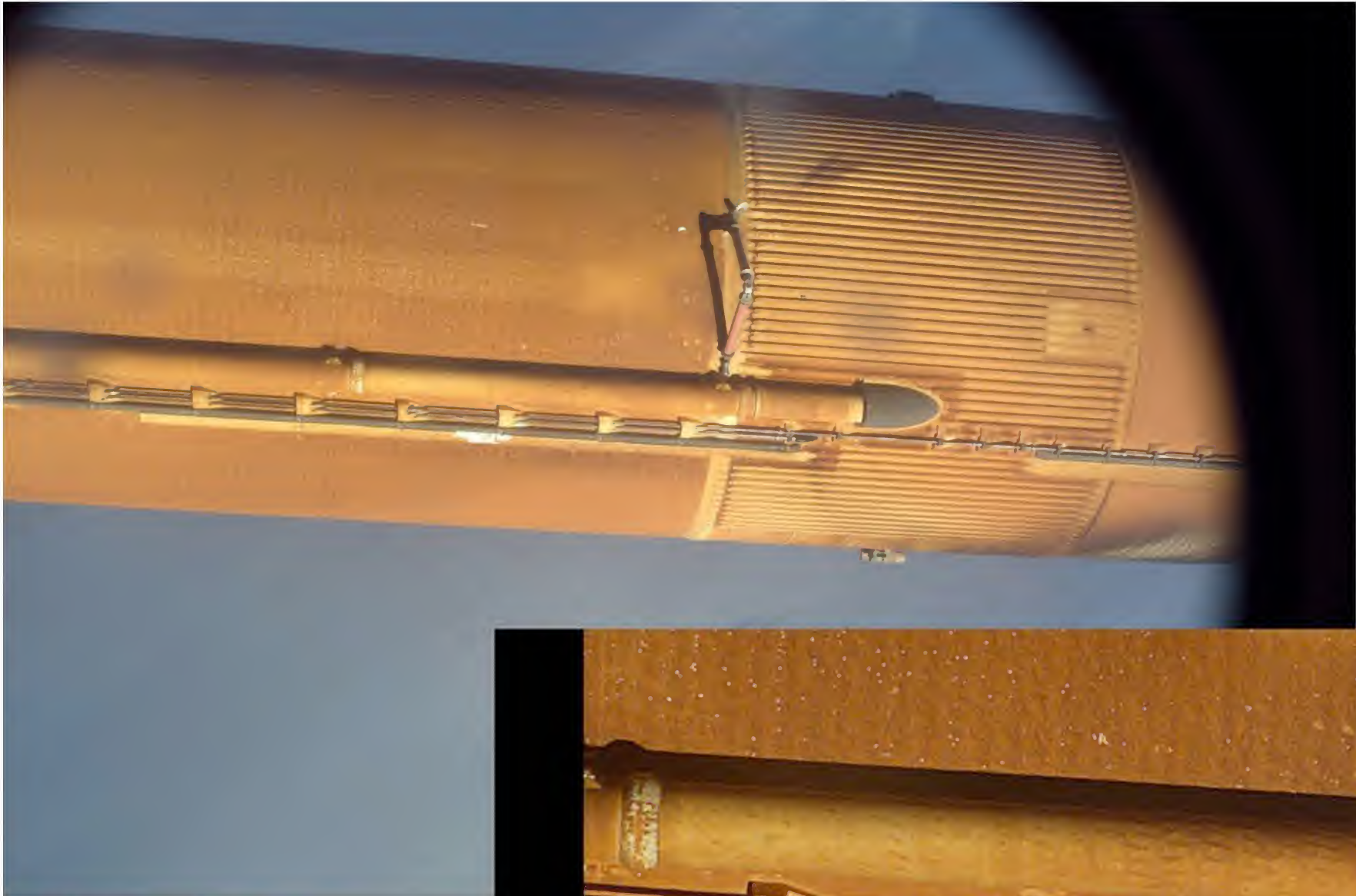
(August 2007)

Still establishing THZ and BSX inspection system capabilities.

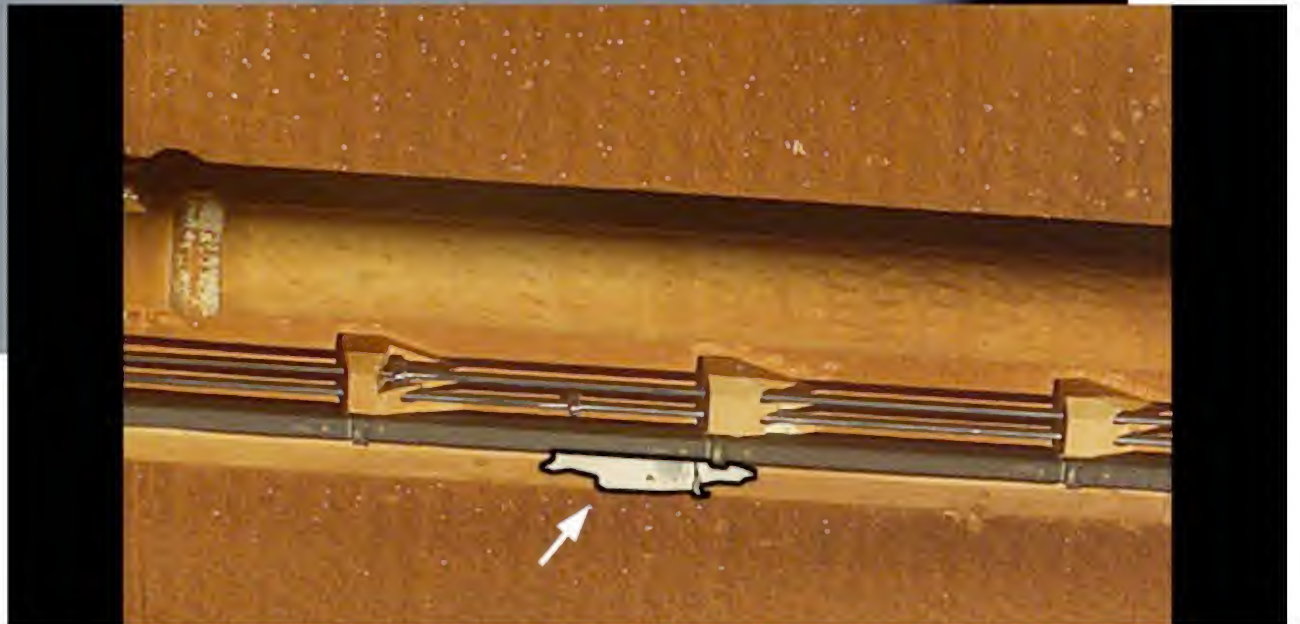




**“We have no hard inspection requirements at this time, that is the Project office and/or Fracture control board have not explicitly stated what size we must find. We have been working toward the self imposed 50% design ultimate critical defect size at a 90/95 level as a point to demonstrate capability.” NDI**



S114E5070



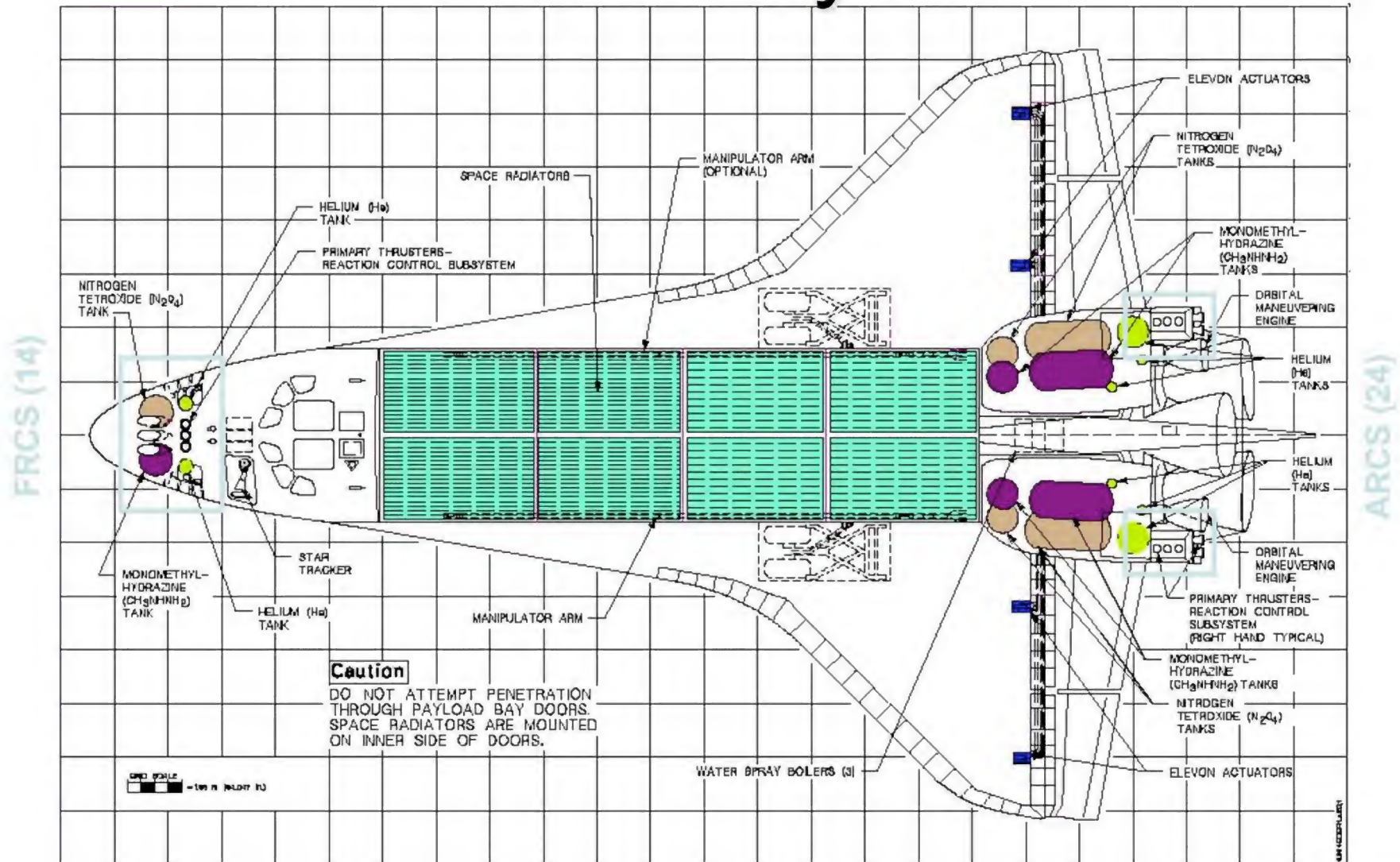


# Primary Reaction Control System Thrusters



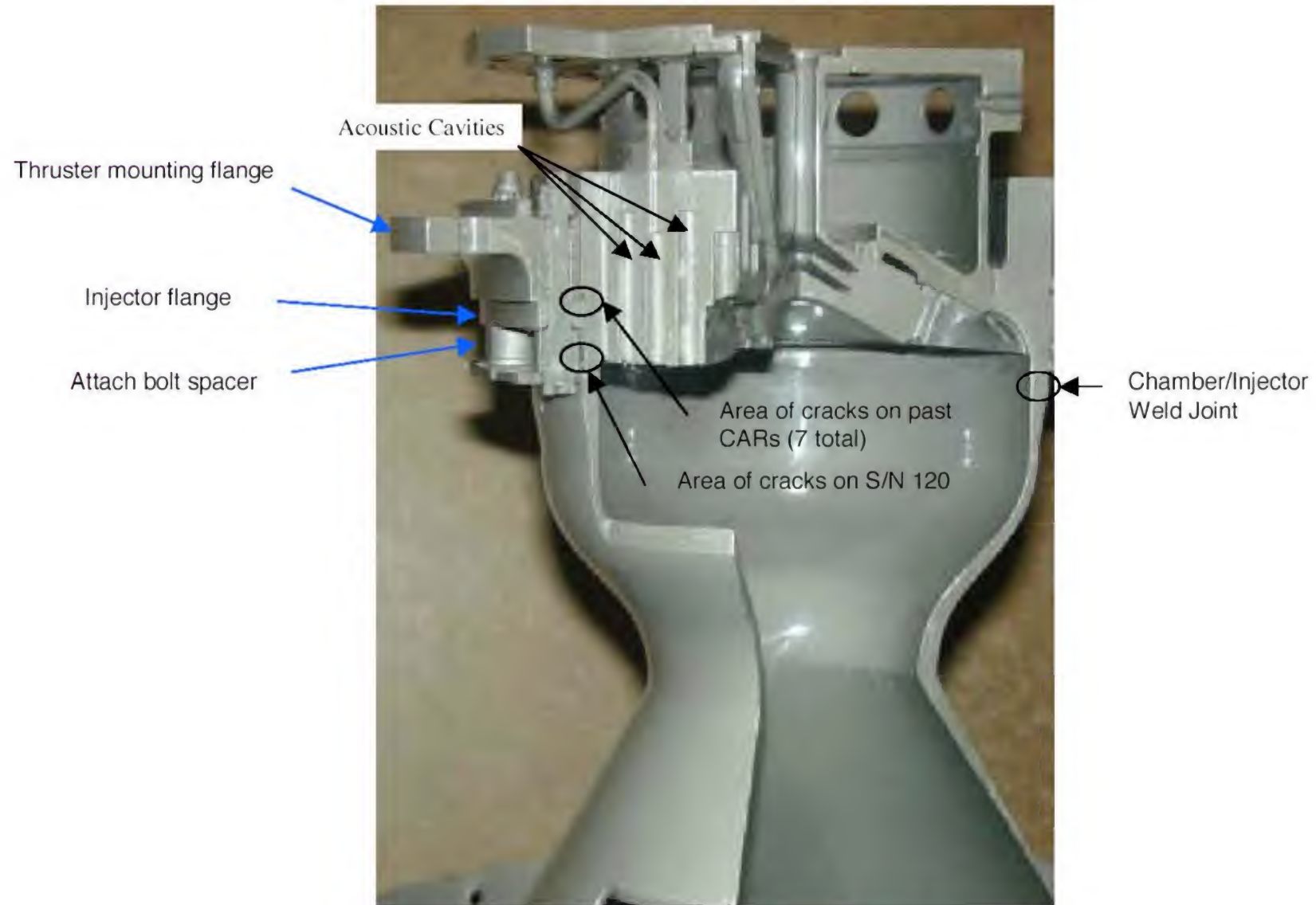


# PRCS Layout



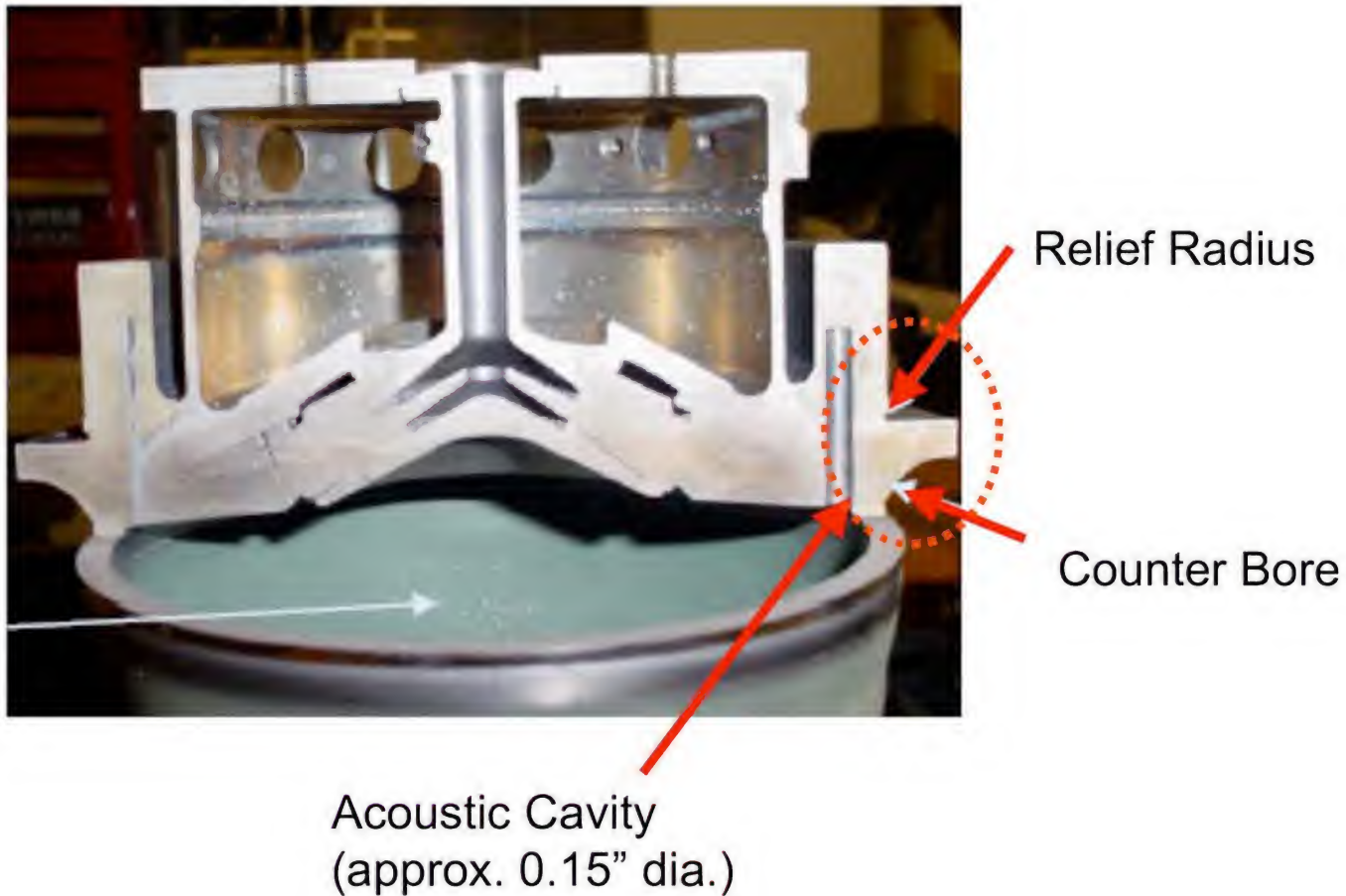


## THRUSTER CRACKS

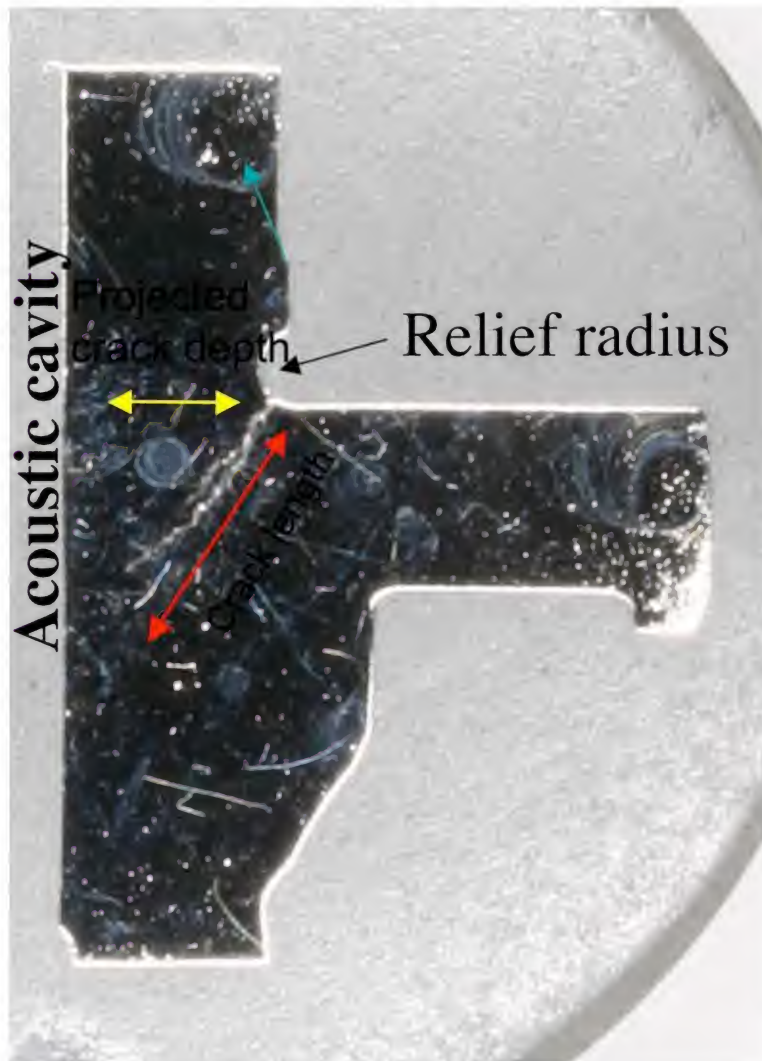




# RCS Injector: Cut Away



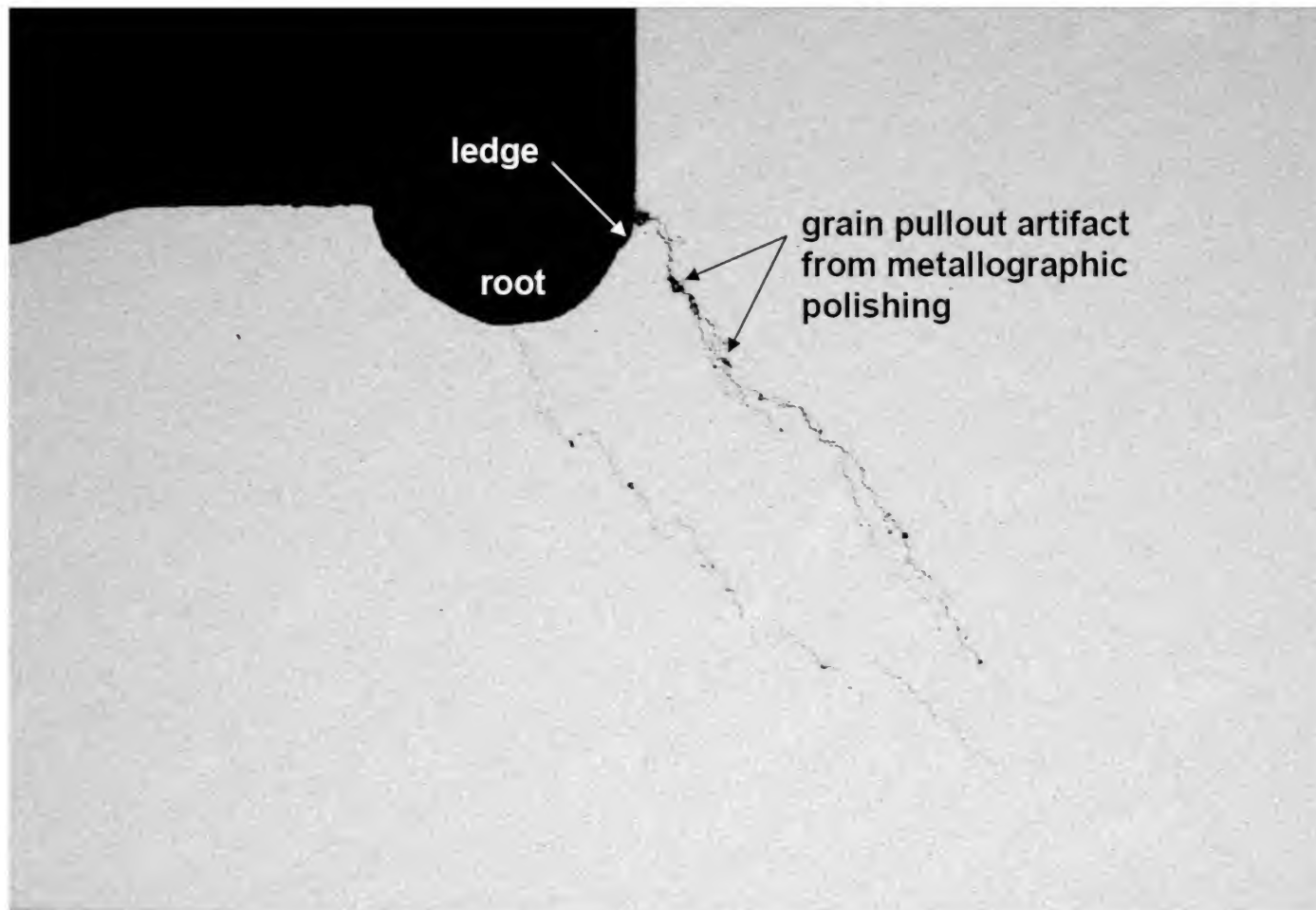




# Metallographic Cross-Section Crack Measurements

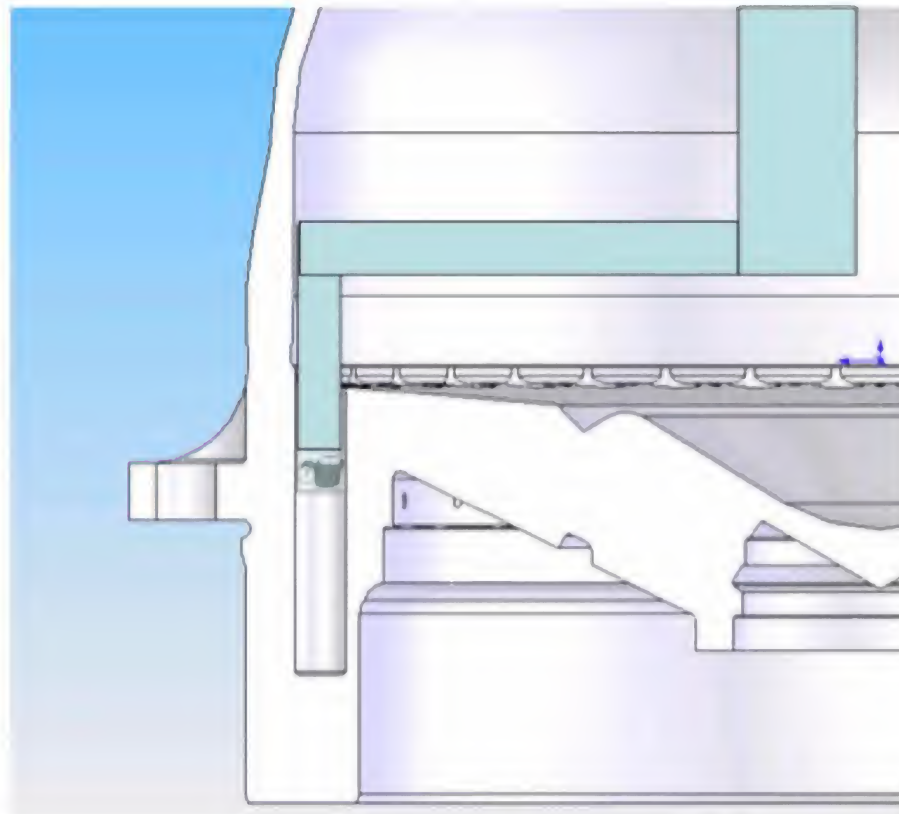


## Multiple Cracks were Observed in Some Polished Sections



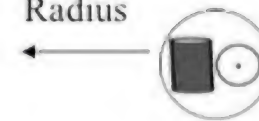


## Orientation of Orthogonal EC Coils Inside Acoustic Cavity for Relief Radius Crack Detection

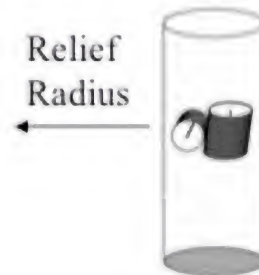


Thruster Cut-away

Relief  
Radius



Relief  
Radius



First Prototype



Projection Drawing

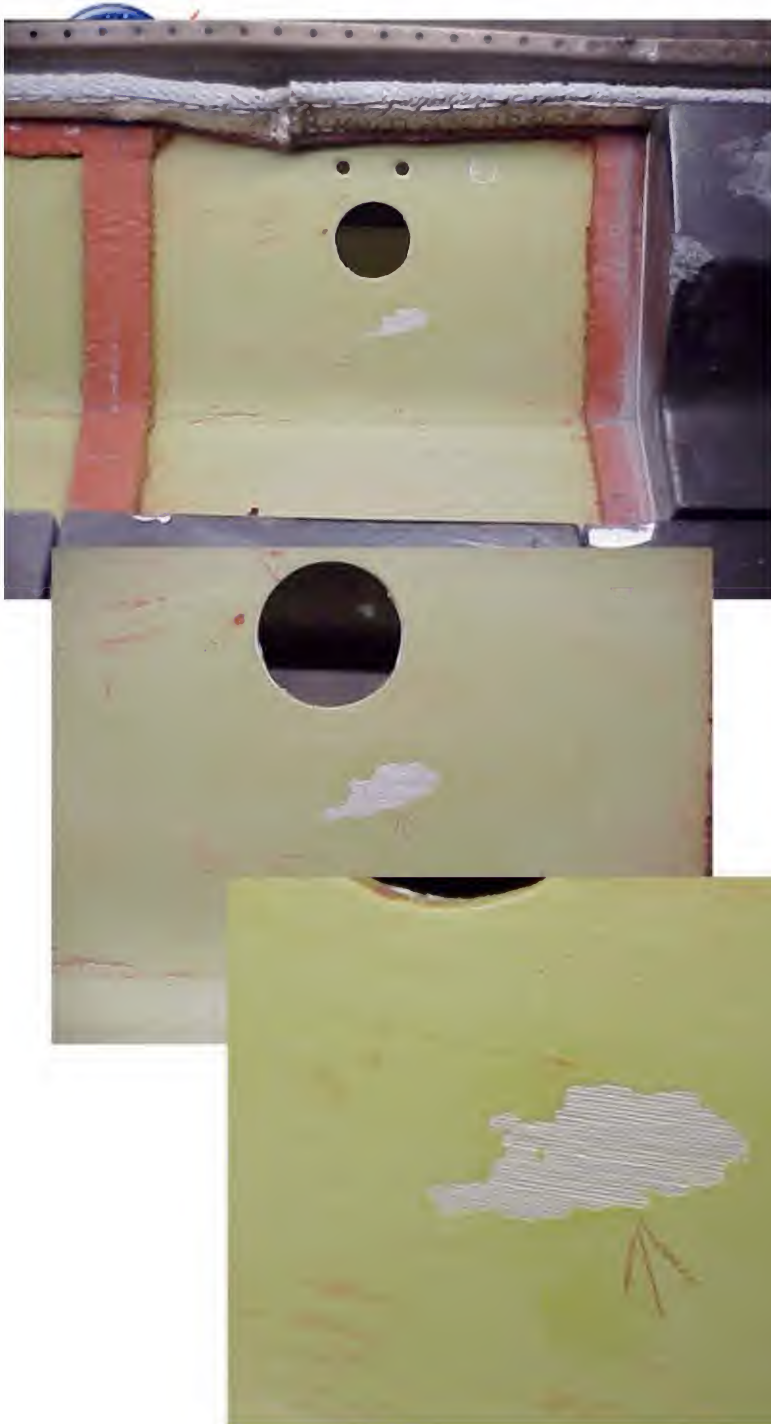




# Corrosion Under Tile

# Corrosion Under Tile

- Corrosion under tile SIP 10/2002
- The Tile/SIP came off with the RTV and koropon attached to it.
- There was a corrosion spot in the middle of the bare aluminum area and it was cleaned up leaving us with what is seen in the third attachment.
- There was no path from the SIP perimeter.





# Composite over-wrapped Pressure Vessels



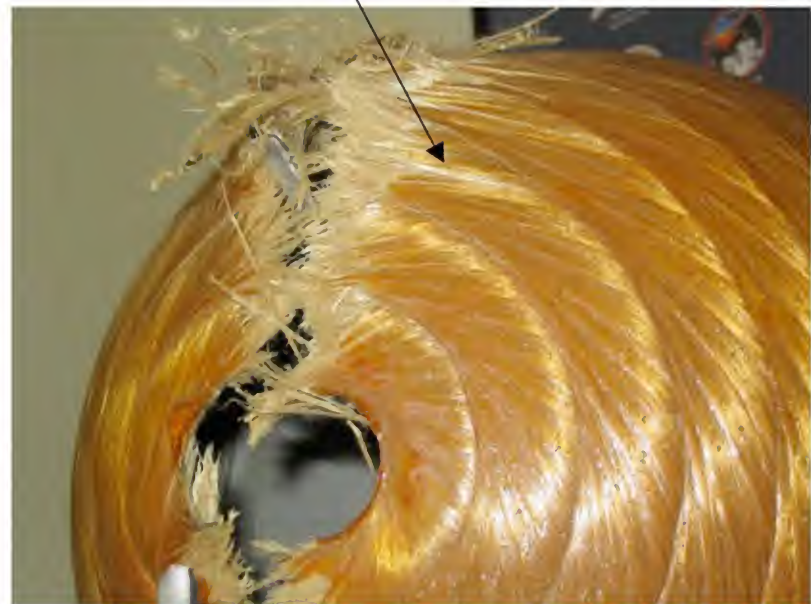


# Composite Overwrapped Pressure Vessels

COPV



COPV failed test article



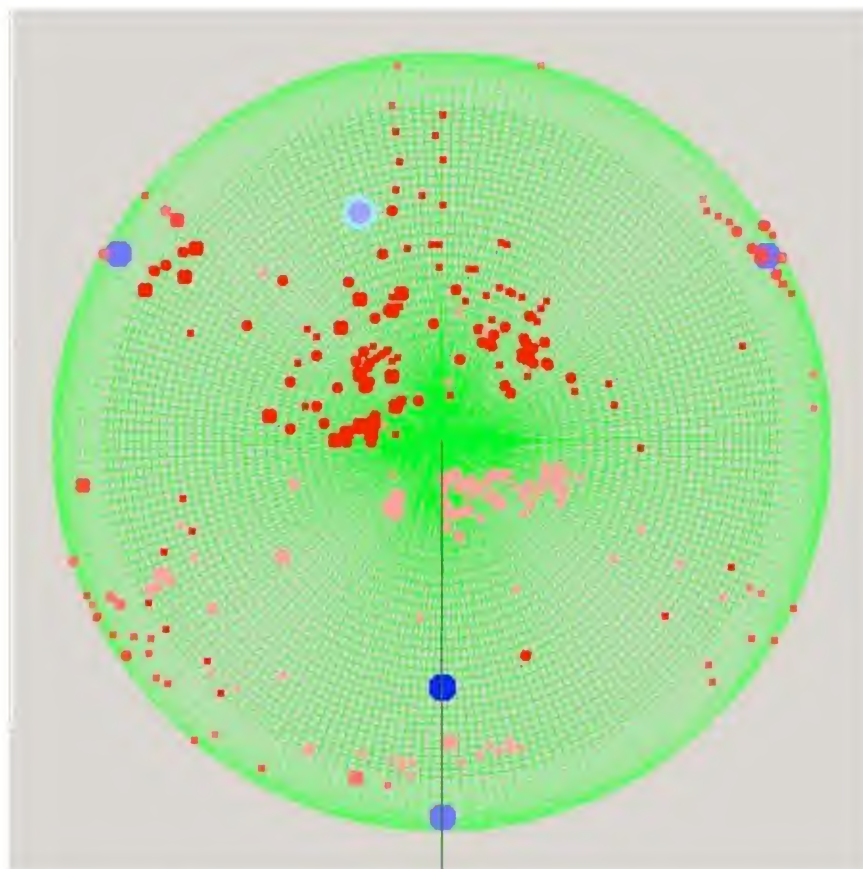


Figure 1. Schematic of Centaur COPV test article showing a cluster of AE events located at the top prior to failure. Blue dots are the location of the AE transducers. Red dots are the location of the AE events.

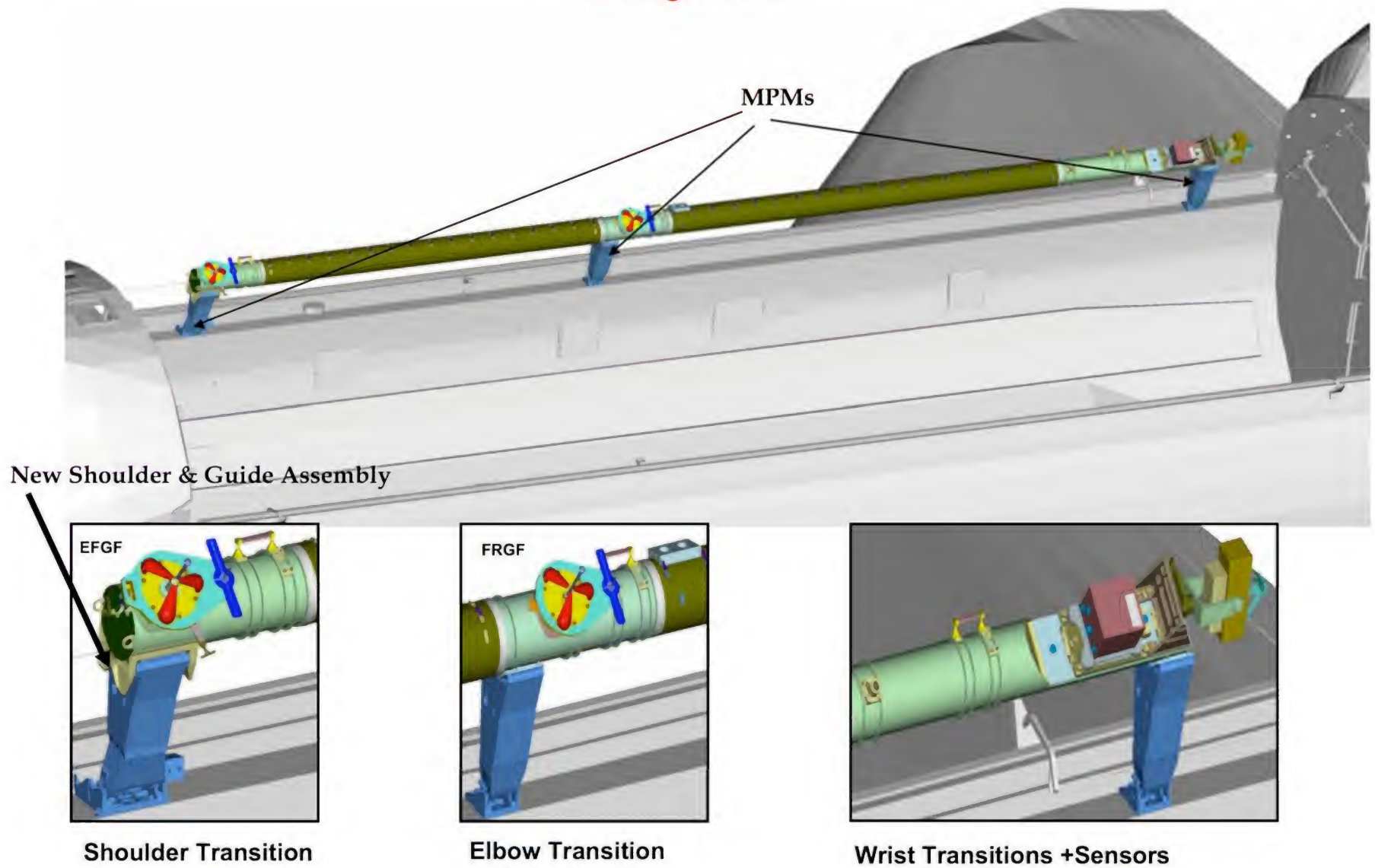


# On-Orbit NDE of RCC



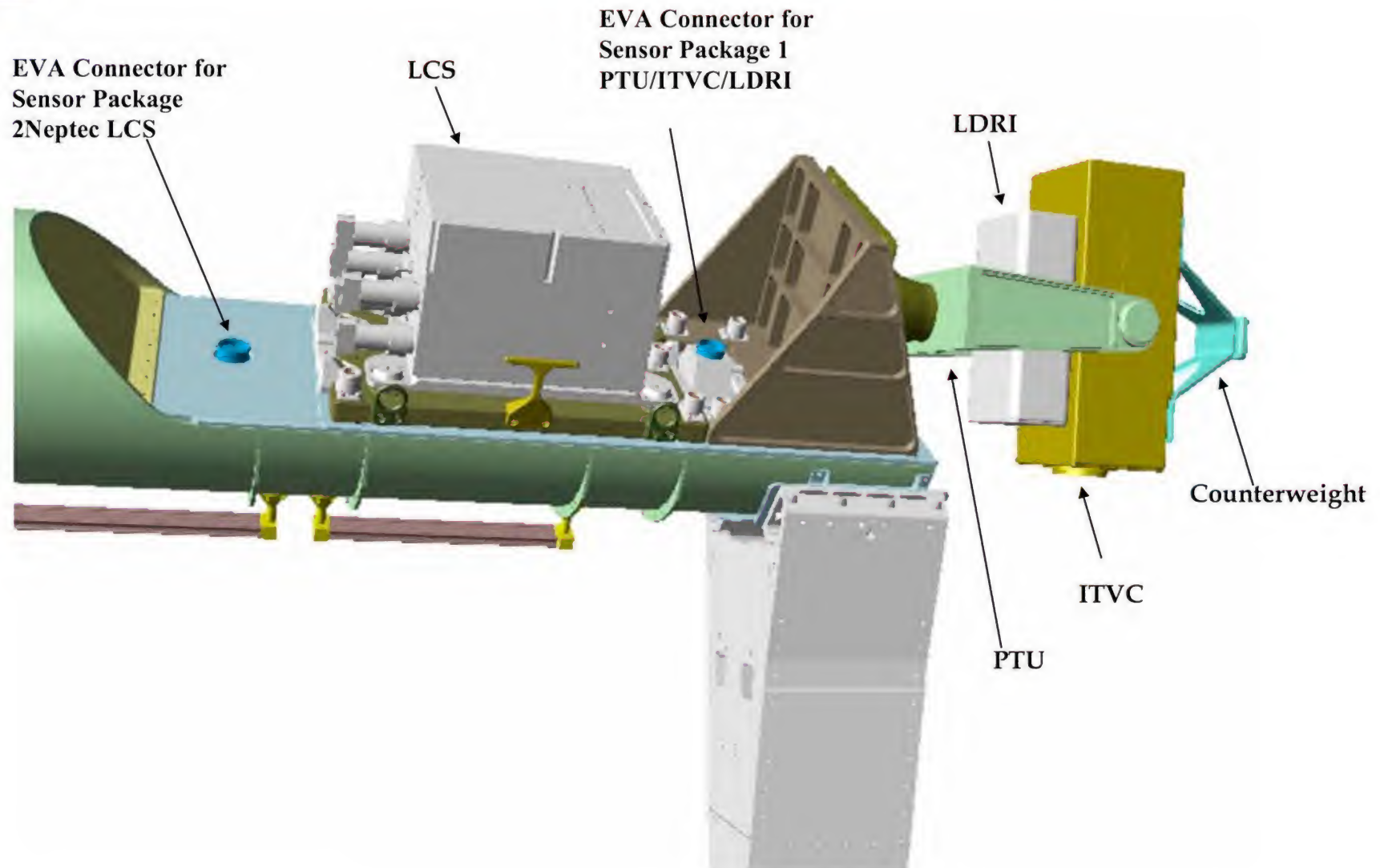


## Background:





## Background: Integrated Sensor Inspection System





# Boom Installed in OV-103







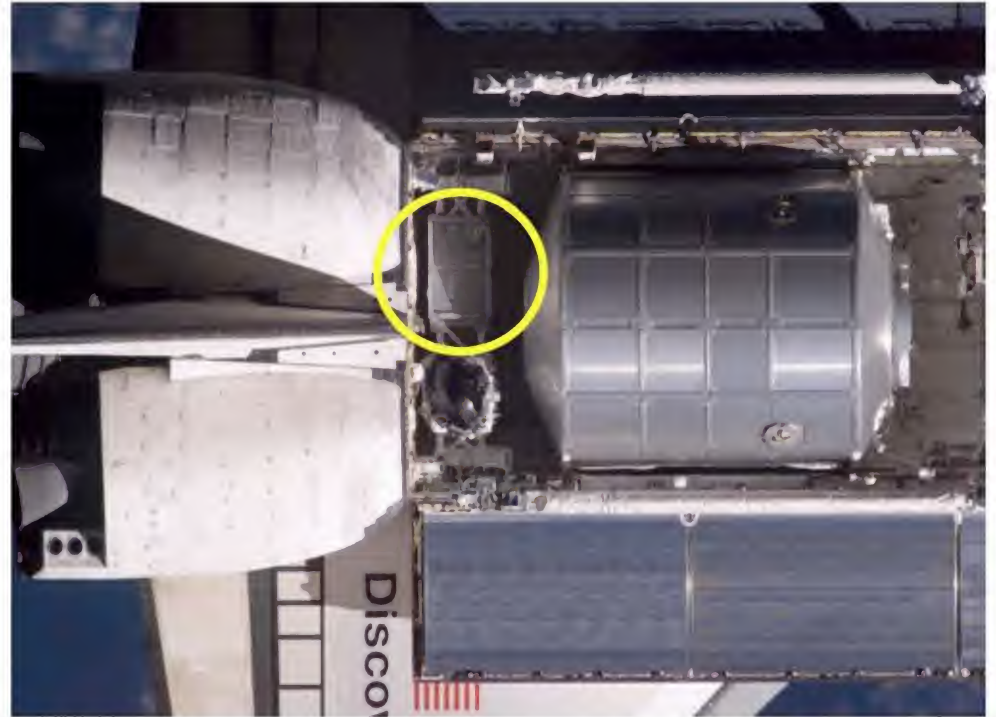
# Inspectors in Space







# Test Samples In Payload Bay

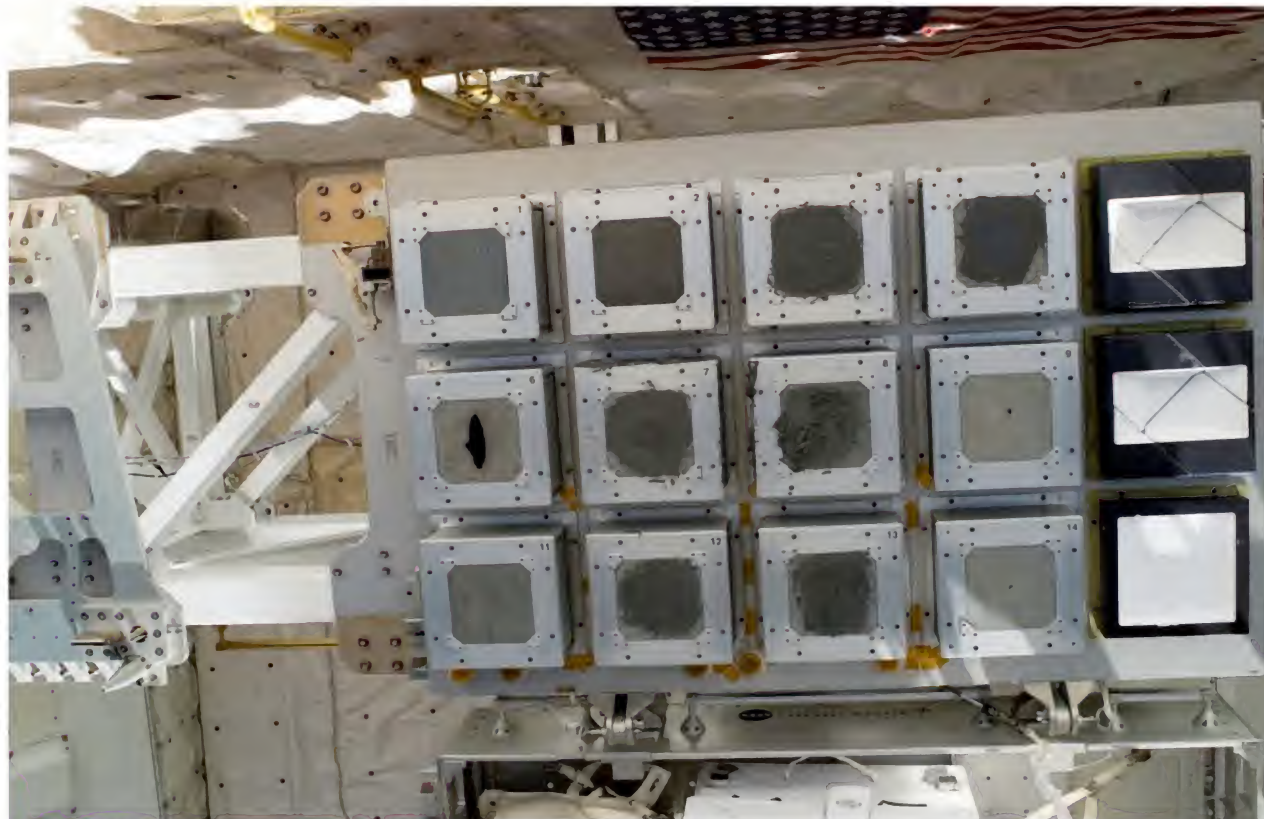


ISS011E11015





# STS-121



S121E06592

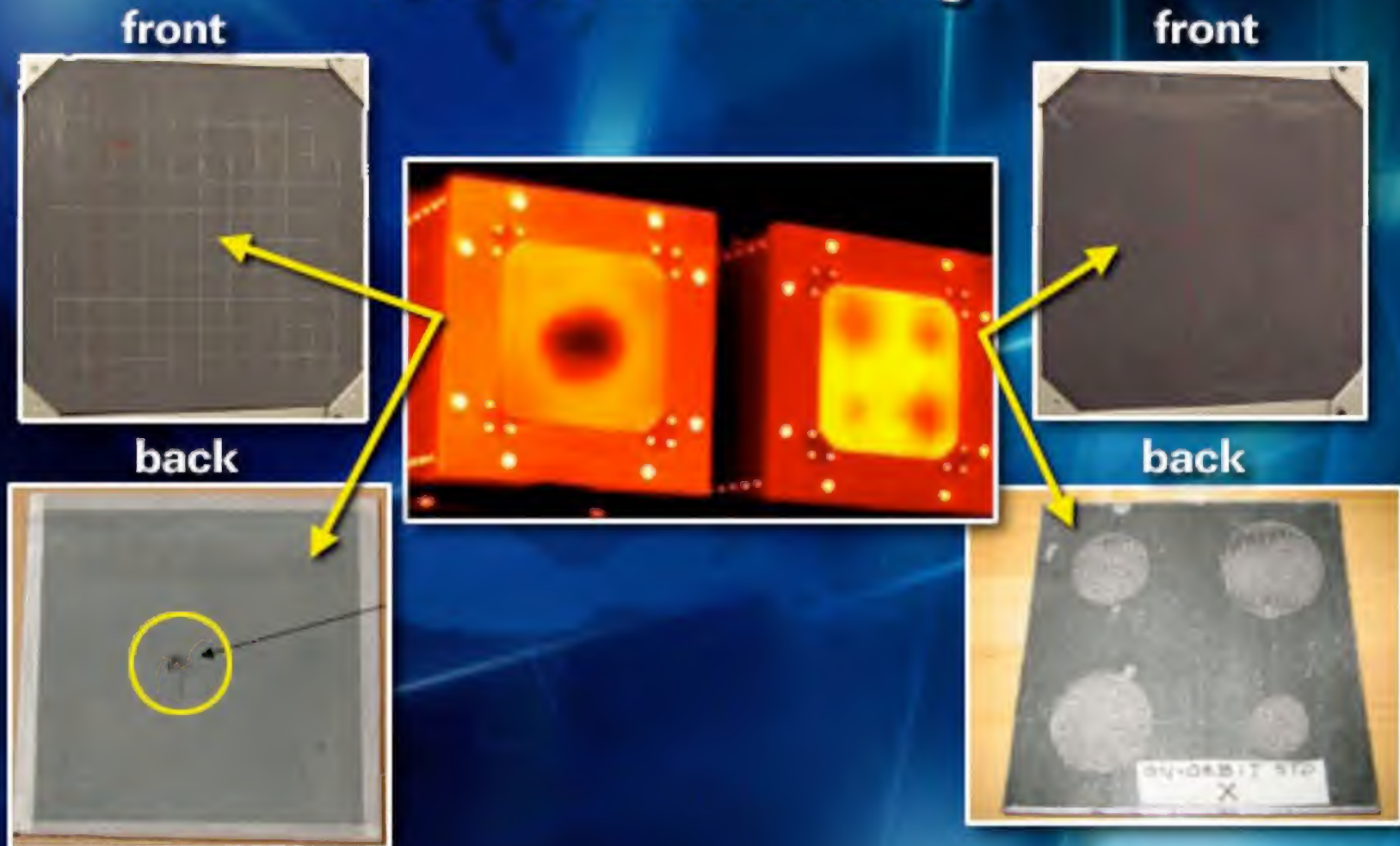


## Inspection In a Relevant Environment

- Inspection Process Movie
- Thermal Data Movie

## Reinforced Carbon-Carbon Samples 1 & 2

### Infrared camera image

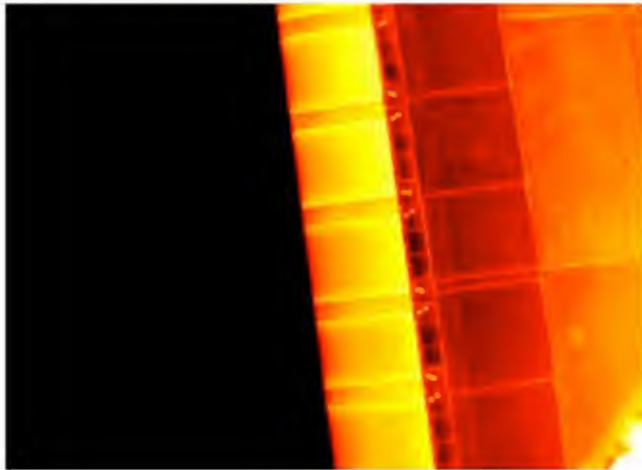






# STS-121 Additional IR Data

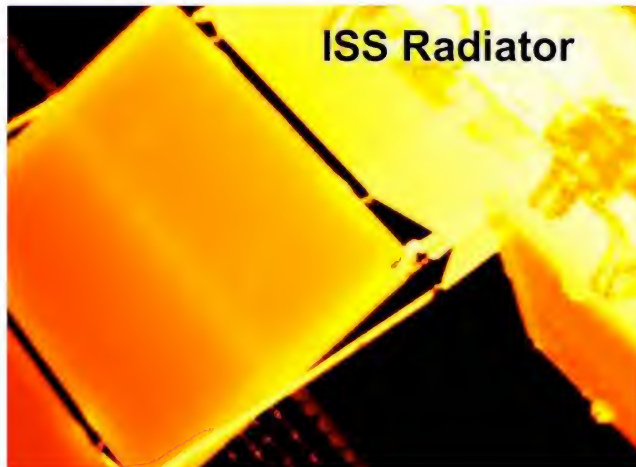
Wing Leading Edge



ISS Radiator



ISS Radiator



Orbiter Radiator





# NASA Standards

**NASA-STD-5009, “Nondestructive Evaluation Requirements for Fracture Critical Metallic Components”**

**NASA-STD-5014, “NDE Implementation Handbook for Fracture Control Programs”**

**NASA-STD-5013, “NASA-STD-5013, “NDE Requirements for Fracture Critical Composite Components”**

**NASA-STD-XXXX, “NDE Requirements for Failure Critical Fitness-for Service Components”, (e.g., non-structural systems like foam, RCC, etc)**



# NASA Standards

## **NASA-STD-5009, “Nondestructive Evaluation Requirements for Fracture Critical Metallic Components”**

- **Additional critical issues that are now addressed specific applicability**
- **NDE planning for life cycle planning**
- **Document and configuration control**
- **Record retention**
- **Procurement requirements**
- **Demonstrating probability of detection (POD) and establishing NDE capability (accepts Mil-HDBK-1823\* when validated by test)**
- **Requirements for acceptance criteria**
- **Organizational responsibilities**
- **Use of similarity**
- **Physical standards**

**MIL-HDBK-1823 - “Nondestructive Evaluation System Reliability Assessment”, 1999**





# NASA Standards

## **NASA-STD-5014, “NDE Implementation Handbook for Fracture Control Programs”**

- Being drafted
- Application to Special NDE Inspector Certification

## **NASA-STD-5013, “NASA-STD-5013, “NDE Requirements for Fracture Critical Composite Components”**

- Initial draft, but has a long way to go
- Supporting ASTM “Standard Guide for NDT of Polymer Matrix Composites Used in Aerospace Applications”, and 5 ASTM Standard Practices documents for composites covering thermography, shearography, radiology/radioscopy, acoustic emissions, and ultrasonics.

## **NASA-STD-XXXX, “NDE Requirements for Failure Critical Fitness-for Service Components”, (e.g., non-structural systems like foam, RCC, etc)**

- Not started



## CEV Crew Exploration Vehicle

The Contractor shall perform 100% non-destructive inspection of all critical components. If the standards listed above are not applicable to specific components, then the contractor shall develop and deliver acceptable (approved by NASA) standard practices and physical standards for non-destructive inspection of components. The Contractor shall develop a Nondestructive Evaluation Plan for Component Acceptance and Life Cycle Maintenance to verify the integrity of delivered components and for the maintenance of components throughout the system's life cycle. The Nondestructive Evaluation Plan for Component Acceptance and Life Cycle Maintenance shall include, as a minimum, elements addressing inspection personnel qualification, inspection processes, qualifying inspection procedures, inspection technologies, records, nondestructive inspection for raw materials acceptance, in-process nondestructive inspection for assembly, acceptance, and final test, critical components list linked to the fracture control plan, and nondestructive inspection for life cycle maintenance. The Nondestructive Evaluation Plan for Component Acceptance and Life Cycle Maintenance shall have configuration control linked to engineering design drawings, critical components lists, and risk analysis documents. Non-critical components shall be subjected to State-of-the-Art non-destructive evaluation practices and requirements. All inspection processes, procedures, data collected, and results shall be made available.



# Probability of Detection (POD)

*"I'm too much of a knucklehead to know this stuff"*

*"Oh boy, confidence limits. I hate these."*

*"They don't use 90/50 they use 90/95."*

*"I should have used 90/50"*

*"I defer my answer to the statistician"*

*"I'm not a statistician."*

*"90/50 POD means that there is a 50% chance that the true POD is greater than 90% at that flaw size?" Responses: "No.", and "Yes.", rest of world gives blank stares.*

*"Confusion over 'common definitions' continues to be an issue..."*

*"We have been using 29 out of 29 clandestinely for years"*





# Probability of Detection (POD)

- A core issue here is that the NDE personnel, nationwide, have different levels of understanding of statistics, and have delegated basic NDE POD statistical analysis to the statisticians.
- NDE community should not blindly accept statistical results, but rather challenge the statistical results.
- When NDE personnel defer explanations on statistical confidence bounds to others, it's like saying "I don't understand the error bars of my data".
- This is not a good position.
- We all need to learn and to speak the language of the other.
- See [DOEPOD presentation later in the week](#)



## 95% Confidence Level Example

### *Lower Confidence Bound, $P_l = 0.9$*

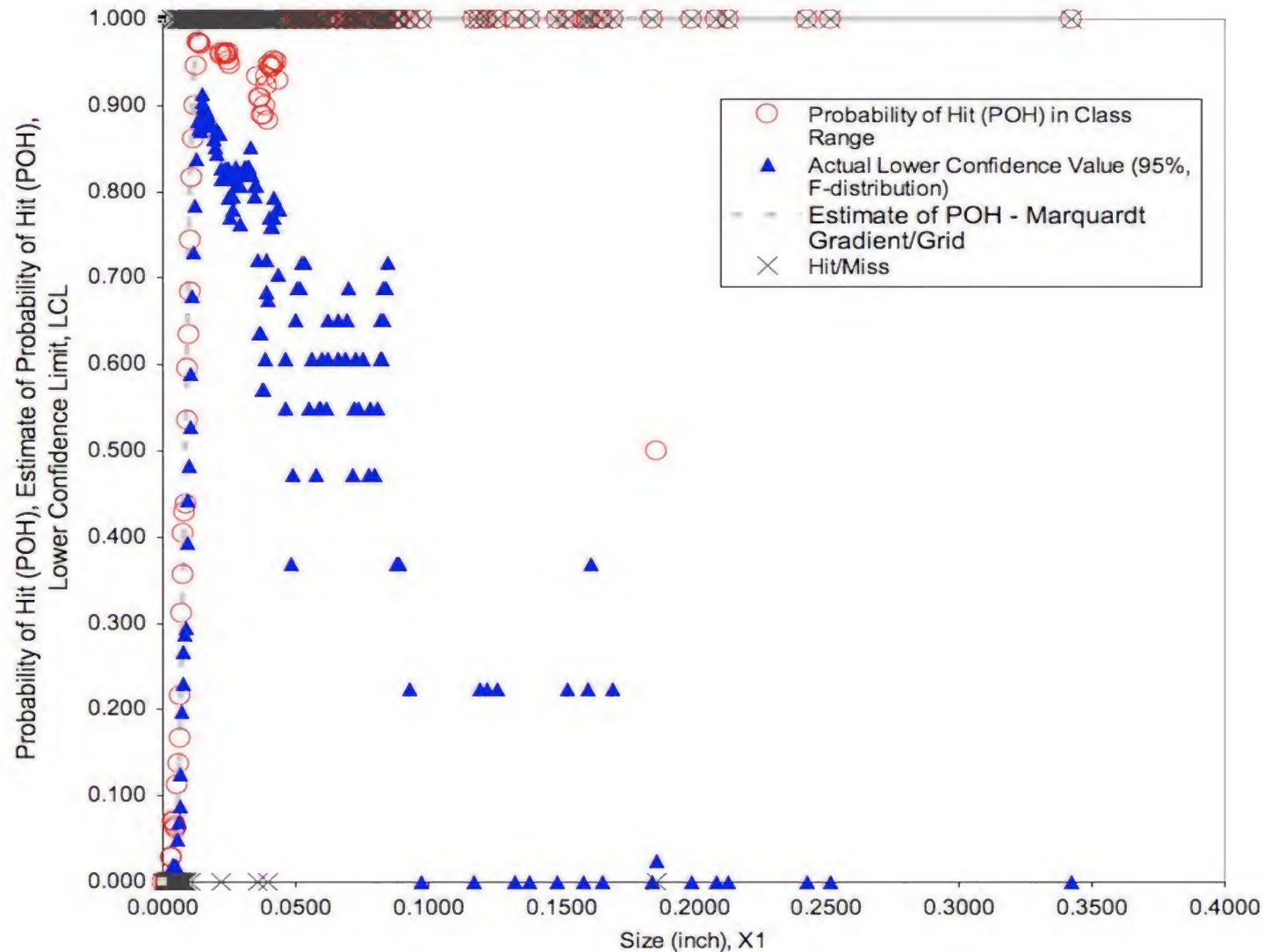
- There is an 95% chance that the true POD is greater than 90% at that flaw size
- There is a 95% chance that the inspection system reliability is greater than 90% at that flaw size
- 90/95 POD at that flaw size

These are all the same



# Design of Experiments for Validating Probability of Detection Capability of Inspection Systems (DOEPOD)

Detection Probability (Utilization of DOEPOD results requires approval of Engineering Authority)







# Looking Toward the Future



## KNOWN MAJOR COMPONENTS WITH ISSUES REQUIRING NDE/HEALTH MONITORING

### Fuselage & Engine Structural Systems

Thermal Protection Systems

Insulation Blankets

Heat Exchangers

Fuselages

Wings

Tails

Platforms

Struts

Debris Shields

Docking Hardware

Seals

Re-entry Systems

Windows

Antennas

Nose Cones

Fuel Casings

Fuel Tanks

Fuel Tank Lines

Feed Lines

Nozzles

Turbopumps

Bearings

Impellers

Housings

Blades

### Electrical Systems

Batteries

Solar Arrays

Tethers



## MONITORING REQUIREMENTS AND REACTION TIMES

### Micro Seconds

Impact Damage  
Pressure  
Leaks

### Seconds

Cracks  
Debonds  
Composite  
    degradation  
Surface  
    contamination  
Mass loss/gain  
Thinning  
Delaminations  
Dielectric  
    breakdown

### Minutes

Alignment  
Misfits  
Joints  
Stress Levels  
Load Verification  
Stiffness  
Dosimetry

### Hours

Thermal  
    Properties  
Humidity  
Temperature  
Chemistry  
Absorbitivity  
Emissivity  
Calorimetry  
Fatigue  
Creep  
Plasticity  
Corrosion  
Erosion

- Time frames generally overlap
- Advanced NDE technologies can address these inspection requirements





# Structural Vehicle Health Monitor & On-Orbit NDE



Long Term Vision

Impact Monitor  
Concept Demonstrator

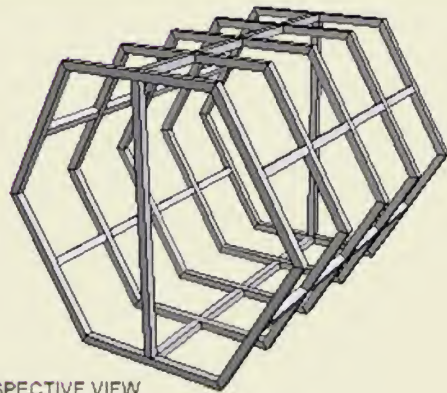
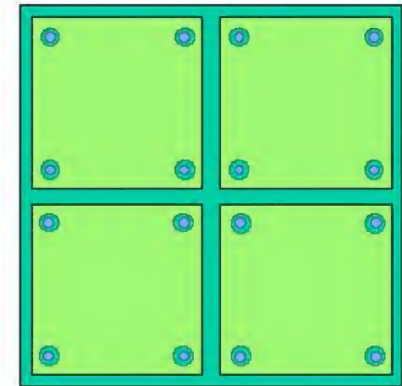
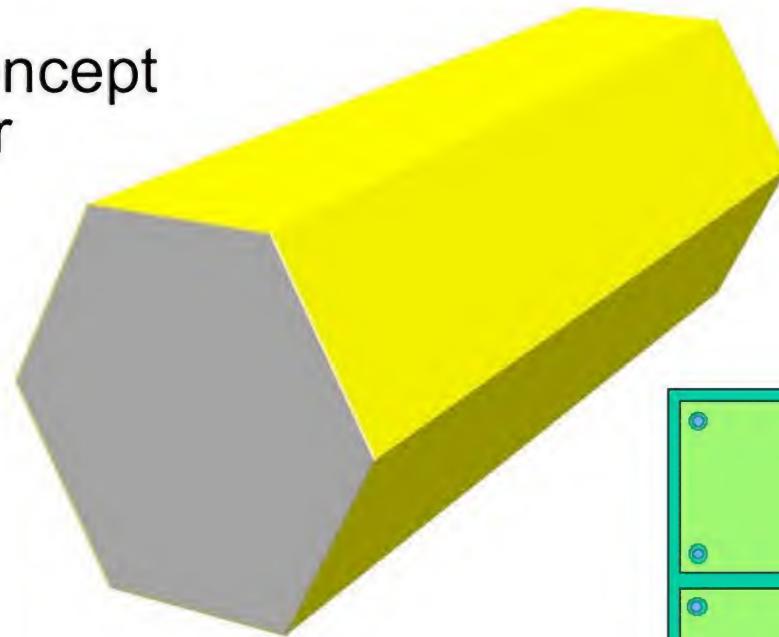


- Structural Integrated Health Monitor
- Companion Astronaut Friendly On-Orbit NDE systems (Thermal, Borescope, etc.)

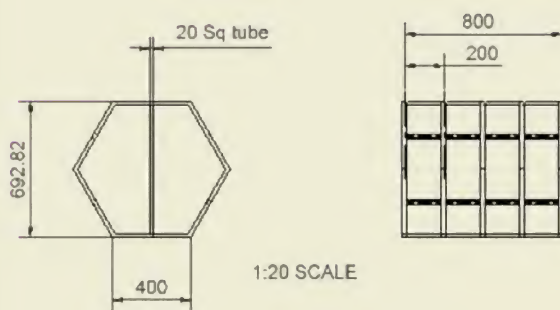


# Impact Monitor Concept Demonstrator

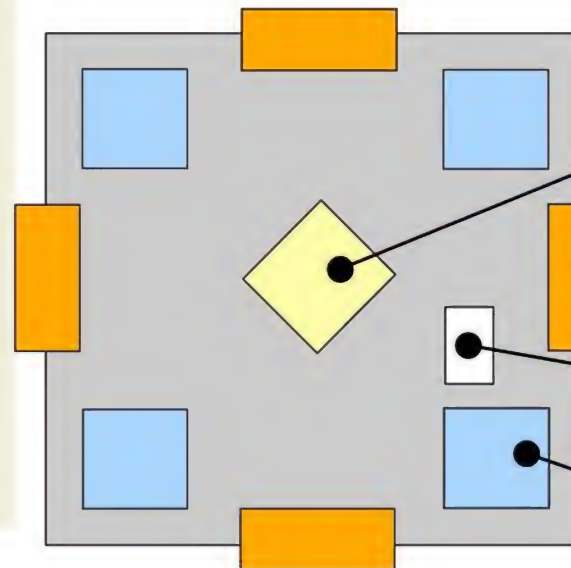
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PERSPECTIVE VIEW



THIRD ANGLE PROJECTION



● LOGIC: PROCESSOR, MEMORY

● COMMUNICATION PORT

● ENERGY UNIT

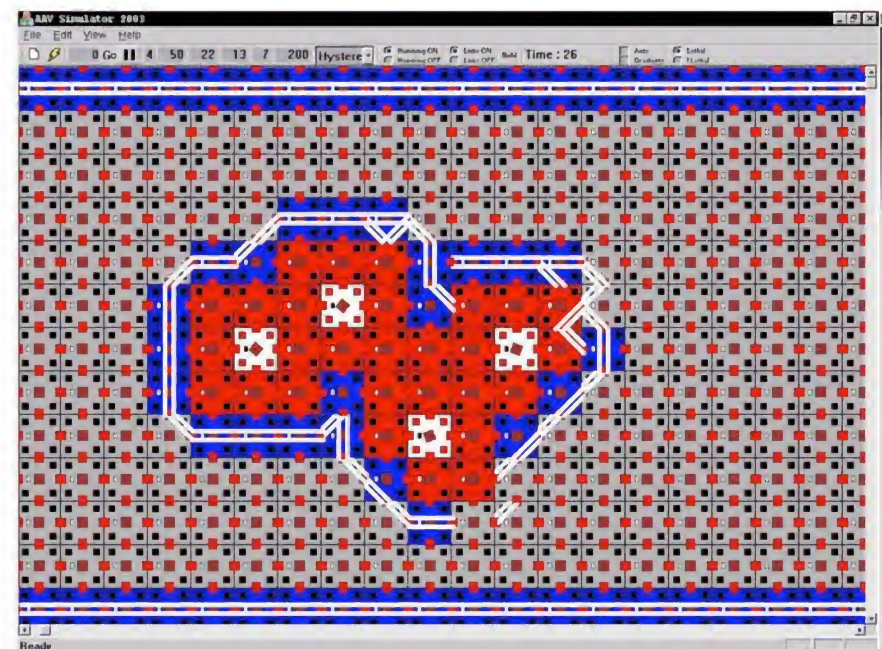
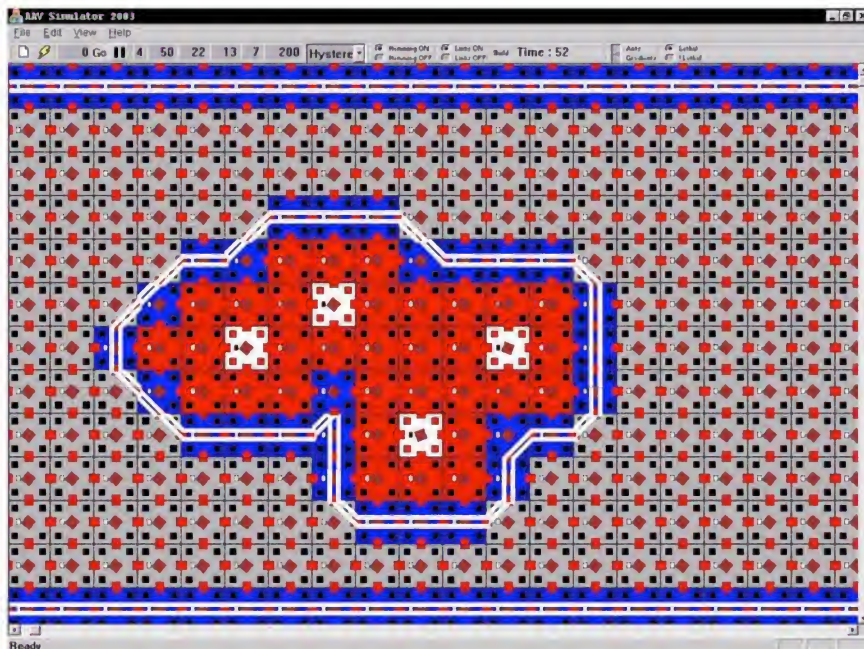
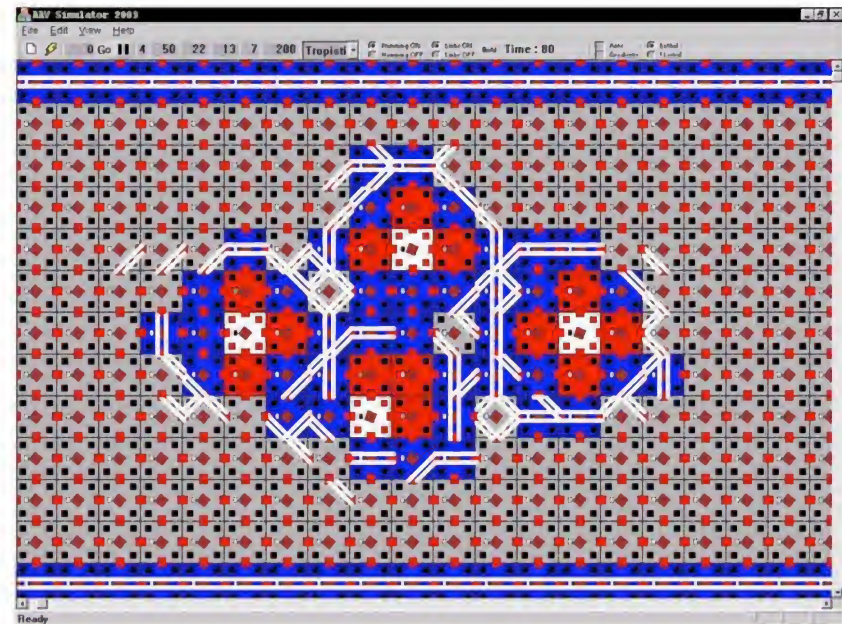
● SENSOR





# Concept Demonstrator

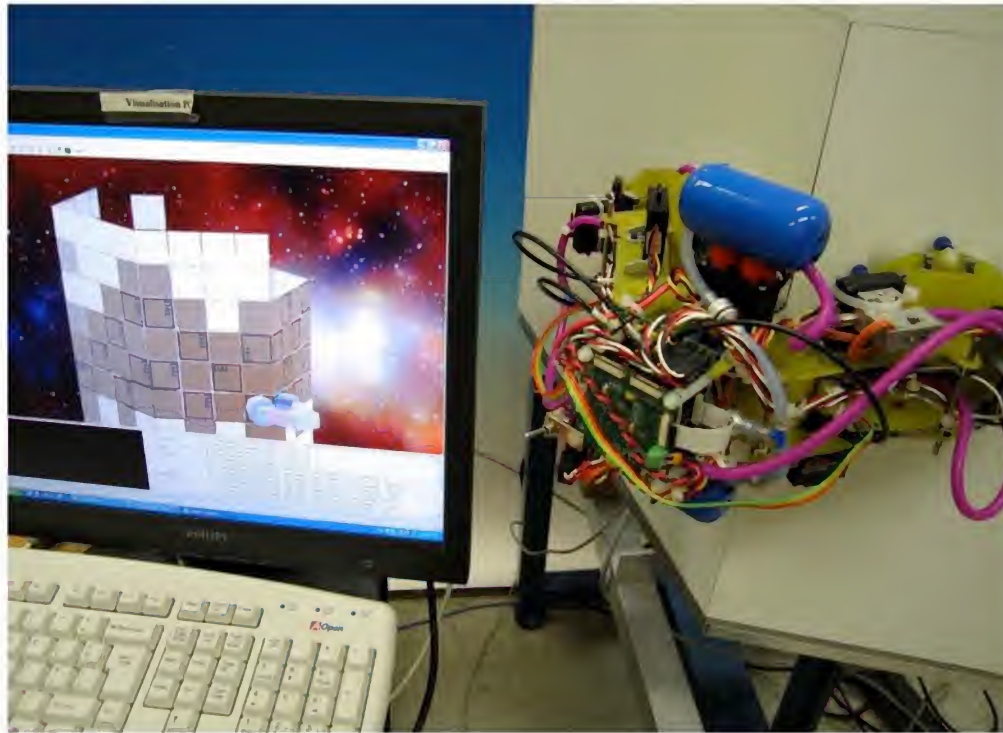
Information and knowledge about impact is maintained by reconnecting damaged communication paths







## Impact Monitor Concept Demonstrator



Visualization screen (left), showing an animation of the NDI robot in its actual position on the CD structure (right).



# Summary

- **Discussed the Columbia Mishap and NDI activities for added mission assurance**
- **Covered projects of the NASA NDE Team & Agency NDE Program**
- **Approaches for Space Transportation System unique inspection issues**
- **Evaluated capabilities for On-Orbit NDT (Inspectors in Space)**
- **Highlighted NASA Core NDT Standards**
- **NDE/Health monitoring approach preparing for the future**